

In the Claims:

Cancel claims 1 and 2 without prejudice or admission.

1. (cancelled)

2. (cancelled)

3. (original) A hardface material composition having improved oxidation and wear resistance at elevated temperatures consisting essentially in weight percent of about:

	Percent
Carbon	0.08 max
Silicon	3.00 - 3.80
Phosphorus	0.03 max
Sulfur	0.03 max
Chromium	16.50 - 18.50
Molybdenum	27.00 - 30.00
Nickel + Iron	3.00 max
Nitrogen	0.07 max
Oxygen	0.05 max
Lanthanum	0.02 - 0.12
Cobalt	remainder

Kindly add the following new claims 4-7.

4. (new) A shroud for an airfoil part of a gas turbine engine, comprising: an interlocking section of a shroud for an airfoil part of a gas turbine engine; a contact area provided at an area of the interlocking section that come in contact with another part of the gas turbine engine, the contact area having a hardface surface, the hardface surface comprising a hardface material composition having improved oxidation and wear resistance at elevated temperature, the hardface material composition being comprised of an alloy having a relatively small lanthanum addition and a relatively large carbon content.

5. (new) A shroud for an airfoil part of a gas turbine engine according to claim 4;
wherein the hardface material composition comprises An alloy characterized by
improved oxidation and wear resistance at elevated temperatures consisting essentially in
weight percent of about:

	Percent
Carbon	0.07 - 1.00
Manganese	1.00
Silicon	1.00
Chromium	26.00 - 30.00
Nickel	4.00 - 6.00
Tungsten	18.00 - 21.00
Boron	.005 - 0.100
Vanadium	0.75 - 1.25
Iron	3.00
Lanthanum	0.02 - 0.12
Cobalt	remainder

6. (new) A shroud for an airfoil part of a gas turbine engine according to claim 4;
wherein the hardface material composition comprises An alloy characterized by
improved oxidation and wear resistance at elevated temperatures consisting essentially in
weight percent of about:

	Percent
Carbon	0.07 - 1.00
Manganese	1.00
Silicon	1.00
Chromium	26.00 - 30.00
Nickel	4.00 - 6.00
Tungsten	18.00 - 21.00
Boron	.005 - 0.100
Vanadium	0.75 - 1.25
Iron	3.00
Lanthanum	0.02 - 0.12
Cobalt	remainder

7. (new) A shroud for an airfoil part of a gas turbine engine according to claim 4;
wherein the hardface material composition comprises An alloy characterized by
improved oxidation and wear resistance at elevated temperatures consisting essentially in
weight percent of about:

Carbon	0.08 max
Silicon	3.00 - 3.80
Phosphorus	0.03 max
Sulfur	0.03 max
Chromium	16.50 - 18.50
Molybdenum	27.00 - 30.00
Nickel + Iron	3.00 max
Nitrogen	0.07 max
Oxygen	0.05 max
Lanthanum	0.02 - 0.12
Cobalt	remainder